

Recent Advances in Near-Net-Shape Fabrication of Al-Li Alloy 2195 for Launch Vehicles

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National Space & Missile Materials Symposium Keystone, CO June 26, 2007



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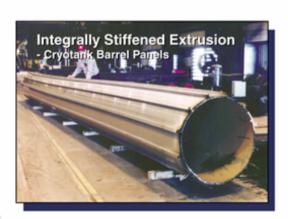
Outline

- Background on Al-Li alloy and Near-Net-Shape Fabrication Technologies for the Launch Vehicle industry
- Ares I Roll Forged Adapter Rings
- Exploration Technology Development Program Spun Formed Domes
- Future Opportunities for 2195 Near-Net-Shape Components
- Remaining Challenges



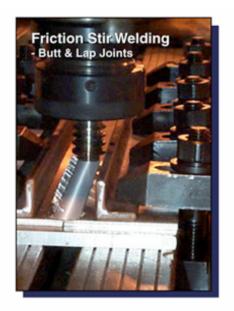
Near-Net-Shape Manufacturing of AI & AI-Li Alloys For Launch Vehicle Structures







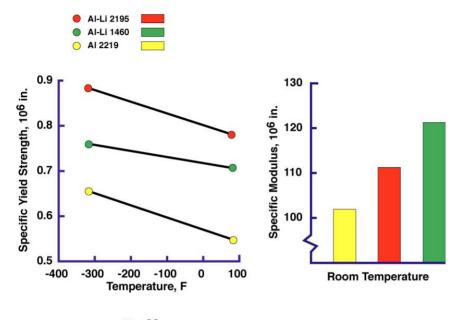


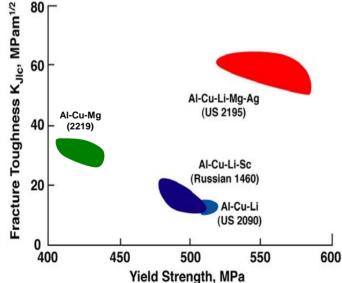






Advanced Aluminum Alloys Technologies for Space Transportation Systems





Benefits of Al-Li Alloys

Lower Density
Higher Strength
Higher Modulus
Good Cryogenic Toughness
Good Formability & Weldability

Current Applications

Space Shuttle External Tank and Intertank Structures
F-16 Bulkheads
Airbus A380 primary structure



Cryogenic Tank Technology Program

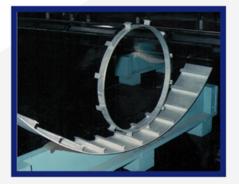


Dome Cap

Adapter Ring

Barrel Section

- Joint NASA LaRC-MSFC-Lockheed Martin
- Applies Al-Li Near Net Forming Methods Developed Through LaRC Programs
- Longitudinal & Circumferential Friction Stirwelds at MSFC
- Concept Demonstration Tank
- Constructed from Al-Li alloy 2195







yogenic tank

Vendors

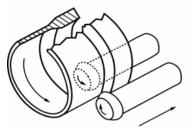
- McCook Metals Co.- 2195 Al-Li Alloy
- Zeppelin Spun Formed Domes
- Wyman Gordon Extruded Barrel Section Panels
- Ladish Corp. Roll Forged Adapter Rings



PROCESS OPTIMIZATION FOR THIN-WALLED 2195 ALUMINUM SHEAR FORMED CYLINDERS



Shear Forming Process





Process Potential

- •Seamless cylinders up to 22 foot diameter
- •Reduce/eliminate longitudinal joints/welds

2195 Shear Formed Cylinder (14' Diameter: 0.18" Wall Thickness)



of 14' Diameter Al-Li 2195 Shear Formed Cylinders

Through-Thickness Microstructures

2195 SFC I

 $t_f=0.15$ inches



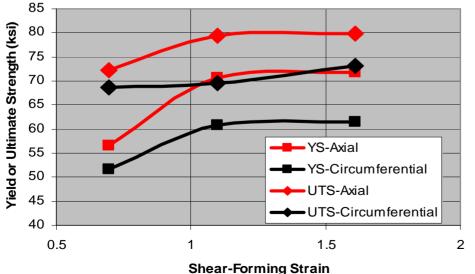
Process parameter optimization yields refined, symmetrical microstructure







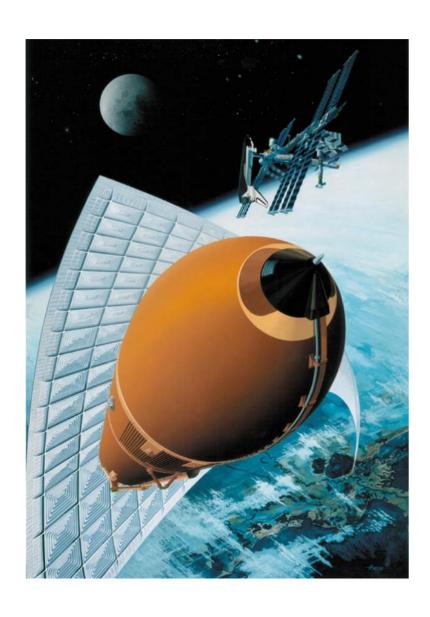
Tensile Properties of 14' Diameter 2195-T8 Shear Formed Cylinder 85



200µm



Space Shuttle Super Lightweight External Tank



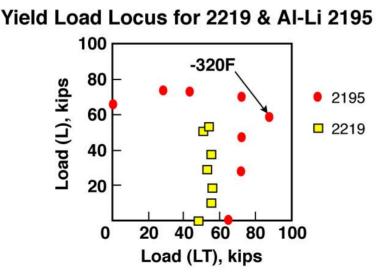
- Space Shuttle system redesigned to support space station launch requirement of 51.6° angle of inclination
- SLWT program requirement was an 8,000 lb. system weight reduction
- System redesigned: T-stiffened → orthogrid
- Material Replacement: Al 2219 → Al-Li 2195
- Program initiated in 1994; First SLWT flight was STS-91, June 1998



Super Lightweight Tank Program Materials Testing and Characterization Program

Biaxial Test Stand





Alloy Development

Composition

Processing

Aging

Plate, extrusions, forgings

Mechanical Property Evaluation

Tensile/Compression

Precision Modulus

Fracture Toughness

Biaxial Tension

Amsler Shear

Metallurgical Characterization

Metallography

Chemical Analysis

Microhardness

Fractography

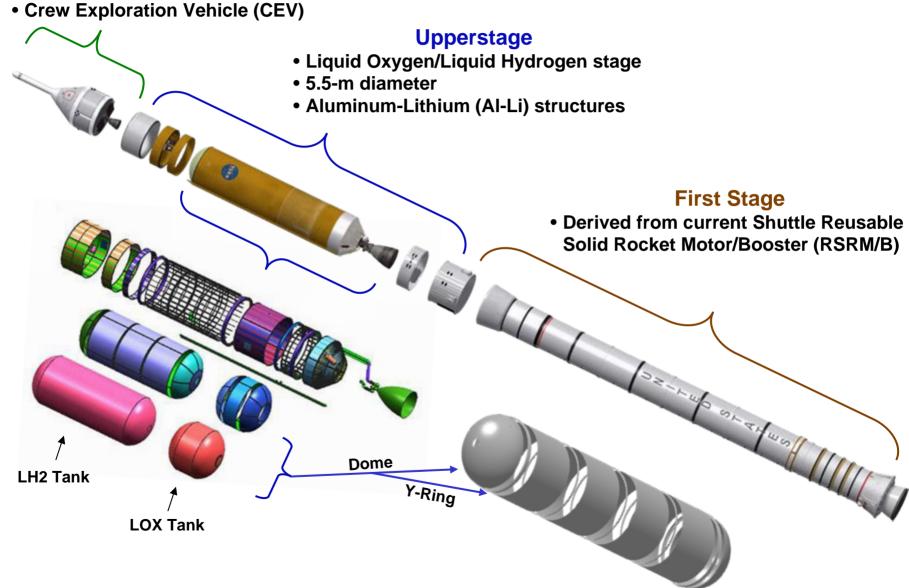
Differential Scanning Calorimetry



Ares I Configuration Crew Launch Vehicle (CLV)



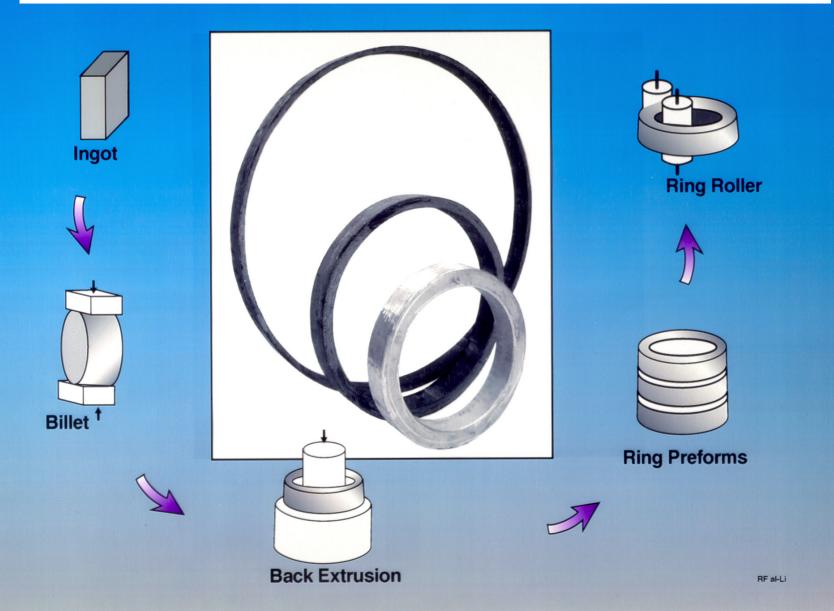
Orion







Roll Forged Al-Li 2195 Process Development Rings



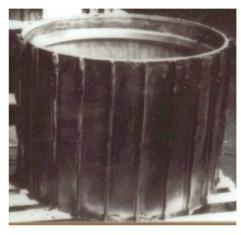




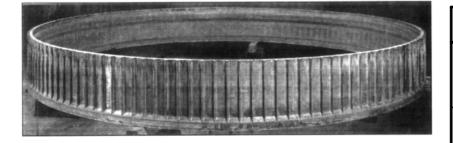
Near-Net-Shape Manufacturing of Al-Li Alloy 2195 for Launch Vehicles

Target Properties								
Temper	Thickness (in)	Orient.	UTS, Min. (ksi)	YS, Min. (ksi)	% el, Min.			
T8A3	0.5 to 2.0 in.	Circ. / Axial	78	73	5			
		Radial	70	63	2			

CLV Upper Stage Manufacturing Requirements Document







Demonstrated Properties								
Temper / Orientation	Temp.	Thickness (in)	UTS (ksi)	YS (ksi)	% el			
T8 /	RT	1.0 in.	83.0	75.0	10.1			
		2.0 in.	82.8	75.6	9.1			
Axial	LN2	1.0 in.	98.4	88.9	7.8			
		2.0 in.	95.6	84.3	10.6			

Average of Multiple Roll Forgings



ARES I Cryogenic Tank Single-Piece Y-ring Adapter Manufacturing Plan





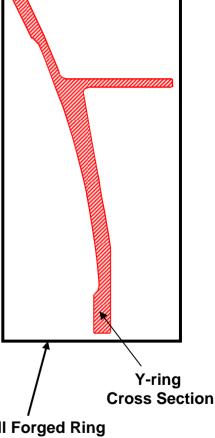
Al-Li 2195 ingot processing to rough pre-form

Roll forge to target diameter Rectangular cross section

Y-ring

Contour machine Post-fabrication > processing

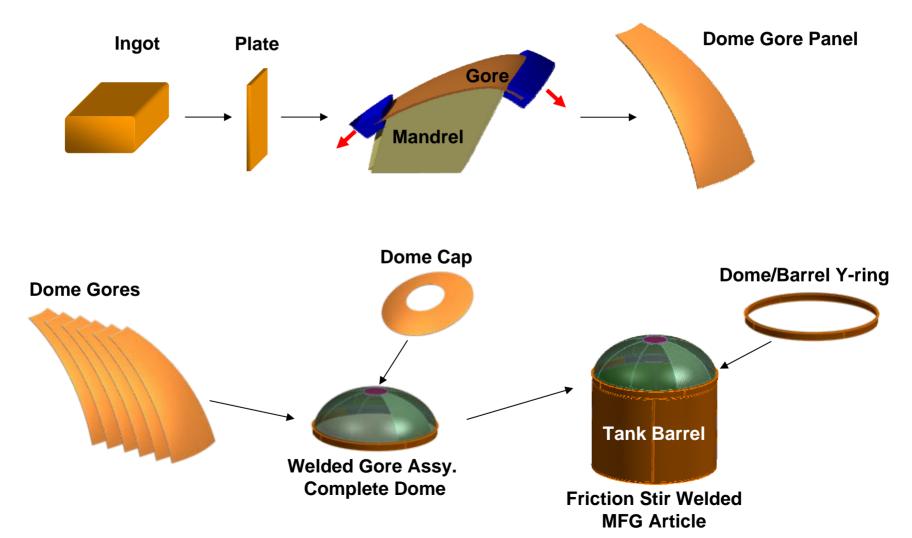
Final Machine



Roll Forged Ring Cross Section



Cryogenic Tank Dome Manufacturing by Gore Assembly Multi-piece; Welded Construction





ARES I Cryogenic Tank Single-Piece Dome Manufacturing Plan



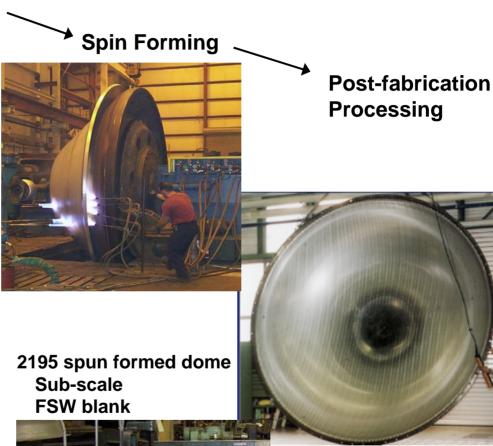
Friction Stir Welded Al-Li 2195 Plate

2195 commercial plate limits spin forming blank size

FSW multi-piece blank

FSW is mature technology for 2195





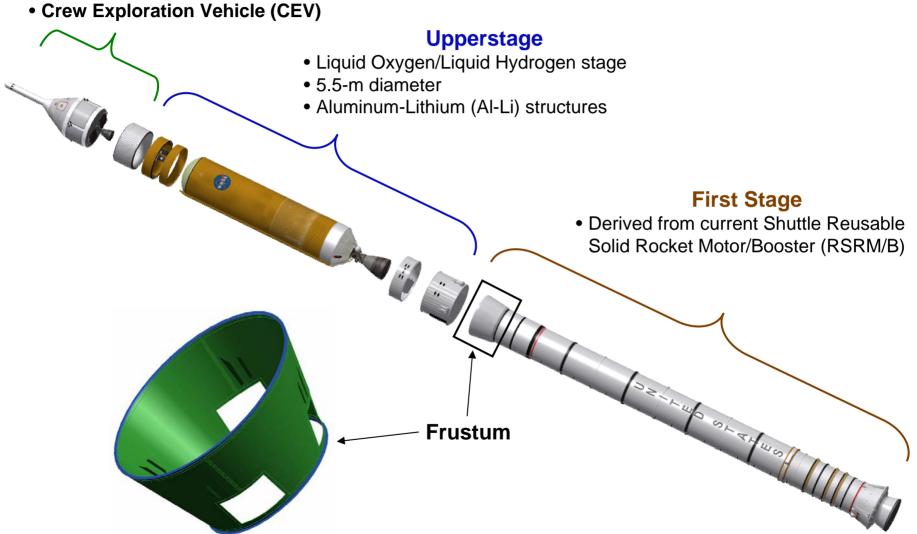
Al-Li Alloy 2195 14-foot diameter Single piece blank



Ares I Configuration Crew Launch Vehicle (CLV)



Orion

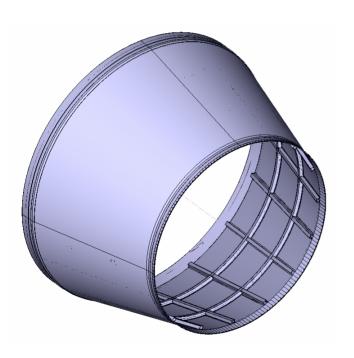




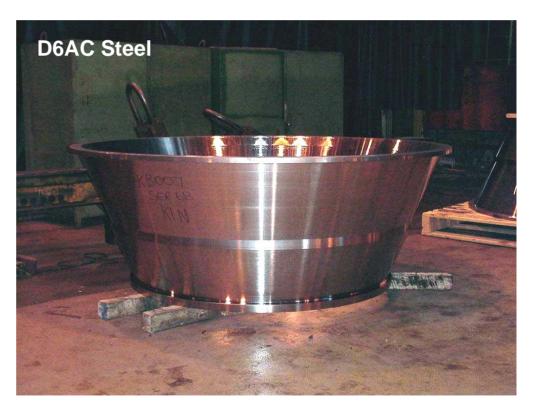
Launch Vehicle Frustum Concept and Ladish Produced Forward Exit Cone



Typical Frustum Structure



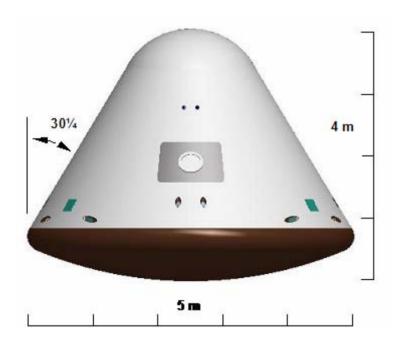
Forward Exit Cone





Potential Applications to Orion Crew Exploration Vehicle (CEV)





- Rings
- Cylinders
- Conical Shapes
- Tanks





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Remaining Challenges

Recent applications in launch vehicles use 2195 processed to Super Lightweight Tank specifications. Potential benefits exist by tailoring heat treatment and other processing parameters to the application.

Assess the potential benefits and advocate application of Al-Li near-net-shape technologies for other launch vehicle structural components.

Work with manufacturing and material producers to optimize Al-Li ingot shape and size for enhanced near-net-shape processing.

Examine time dependent properties of 2195 critical for reusable applications.